Simulation Research on Gravity-Geomagnetism Combined Aided Underwater Navigation

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Gravity Aided Navigation (GravAN) and Geomagnetism Aided Navigation (GeomAN) are two methods for correcting Inertial Navigation System (INS) errors of Autonomous Underwater Vehicles (AUVs) without compromising the AUV mission. One requirement for applying these methods is the relatively large field feature variations along the navigation trajectory. But in some regions with small gravity or geomagnetic variation, it is very difficult to achieve a reliable result solely by GravAN or GeomAN. If these two methods were combined, gravity and geomagnetism information could be complementary and the aided navigation ability could potentially be improved, especially in those regions where neither method is valid. Based on that concept, a Gravity and Geomagnetism Combined Aided Navigation (GGCAN) method is consequently proposed in this paper as a possible solution. The Gravity Anomaly Grid (GAG2) and Earth Geomagnetic Anomaly Grid (EMAG2) are utilized as the background databases, and then a Multiple Model Adaptive Estimation (MMAE) is adopted to obtain an optimal estimated navigation position. Furthermore, an Optimal Weight Allocation Principle (OWAP) is introduced to the combined GravAN and GeomAN methods, together with a weighted average. In simulation, two special regions in the Western Pacific Ocean were chosen to test the proposed method. The results show that GGCAN can improve the position success rate and reduce the error, compared to GravAN or GeomAN. Results indicate that the GGCAN method proposed in this study is able to improve the accuracy and reliability of an aided navigation system.

KEY WORDS
1. Underwater navigation.
2. Aided navigation.
3. Geomagnetic anomaly.

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accumulation. Contrary to the situation on the water surface, where satellite and some other radio navigation signals are easily available, the problem of underwater navigation is more complicated due to the requirement of continuous underwater operation for long periods. Sometimes it is unacceptable for the AUV to acquire the navigation updates by surfacing or nearly surfacing (Wu et al., 2010; Hays et al., 2002; Satow, 1951). However, gravimetry and magnetometry are totally non-emanating and passive, they do not need the AUV to surface. Gravity and geomagnetic fields vary from place to place around the Earth’s surface and this is recognizable by matching algorithms (Liu et al., 2007; Huang et al., 2010). Thus gravity and geomagnetic information can be employed to construct the passive aided navigation systems for AUV.

The design of appropriate and accurate navigation algorithms is one of the most crucial parts in gravity/geomagnetic aided navigation systems. Presently, the most frequently used algorithms based on map matching technology are Terrain Contour Matching (TERCOM) and Iterated Closest Contour Point (ICCP) (Deng et al., 2010). Although both methods are able to achieve credible matching results, a sequence of gravity or geomagnetic measurements that are collected in advance is a necessary condition, so the matching process is by no means simultaneous (Zhao et al., 2009). In addition to map matching technology, an Extended Kalman Filter (EKF) is another choice to obtain an estimated position in underwater aided navigation. It handles the issue under nonlinear filter theory and can estimate the INS navigation error in real time. However, EKF is very sensitive to initial and process errors; small errors may lead to divergence, so the reliability and stability of this method cannot be ensured (Wu et al., 2011).

The focus of previous researches has been on Gravity Aided Navigation (GravAN) or Geomagnetism Aided Navigation (GeomAN) individually (Feng et al., 2008; Hays et al., 2002). The aided navigation result is much related to the gravity or geomagnetic features along the trajectory. In some regions of insufficient gravity or geomagnetism variation, if these two methods were combined together better navigation performance could be expected. Consequently, the Gravity and Geomagnetic Combined Aided Navigation (GGCAN) is proposed in this paper. There are two main procedures in GGCAN. First, Multiple Model Adaptive Estimation (MMAE) is employed as a navigation algorithm, in which multiple Kalman filters are formed and the optimal navigation positions are estimated separately from GravAN and GeomAN. Secondly, the results from GravAN and GeomAN are combined with the weighted average method derived from the Optimal Weight Allocation Principle (OWAP) and the final GGCAN position result can be obtained.

Weighted average methods were extensively used in multi-sensor data fusion, information retrieval and decision making (Hall and Llinas, 2001; Larsen, 2011). With simple weighted average methods like Equal Weighted (EW), the accuracy of fusion result may be less than that of one of the inputs. Thus a better weighted method is needed so that the accuracy of the GGCAN result can be better than either of GravAN and GeomAN. A weighted average method based on OWAP is explored in this study. This method is more effective and convenient. Moreover, the accuracy difference between GravAN and GeomAN is fully considered.

Two regions in the West Pacific Ocean were selected for the simulation and the performance and reliability of GGCAN is discussed. The improvement in the success rate and position accuracy show that OWAP is more appropriate for GGCAN than
the equal weighted method, and GGCAN (OWAP) presented in this study is more accurate and robust than GravAN or GeomAN.

2. PRINCIPLES OF GGCAN. In GravAN and GeomAN, the gravity and geomagnetic data are collected point by point with onboard sensors, as the AUV moves. After pre-processing and calibrations, the measured anomalies are matched with the background anomaly prepared previously, and the optimal estimated positions are obtained with MMAE. Thereafter, the estimated positions are used to update the INS navigation information. In the proposed GGCAN method, both gravity anomaly and geomagnetic anomaly are involved. The GGCAN system consists of INS, aided navigation background databases, gravity/geomagnetic sensors, aided navigation algorithms (MMAE and OWAP), and a navigation processor. A block diagram of GGCAN is presented in Figure 1. The measured gravity/geomagnetic anomalies and background databases are sent to MMAE modules. Multi-module Kalman filters are employed here in the MMAE technology to ensure that reliable position estimations are obtained from GravAN or GeomAN. Then a weight function that is derived from the OWAP is adopted to compute the final GGCAN combined positions from the results of GravAN and GeomAN. Finally, the combined positions are used to modify the errors of the INS.

3. GRAVAN AND GEOMAN ALGORITHM.

3.1. Gravity Anomaly Grid Map and Geomagnetic Anomaly Grid Map. Gravity and geomagnetic background databases are the most basic and fundamental elements of the GGCAN system. In this research, a gravity anomaly grid map and geomagnetic anomaly grid map are selected as the databases.
With the appearance and development of satellite altimetry, the resolution of the Gravity Anomaly Grid (GAG2) in ocean area from altimetry inversion has already reached $2' \times 2'$. Therefore, a KMS2002 gravity anomaly grid map with $2' \times 2'$ resolution is chosen for GGCAN in this paper (Andersen et al., 2005).

The Earth Magnetic Anomaly Grid (EMAG2) was released by National Geographic Data Center (NGDC) in March 2009; EMAG2 is compiled from satellite, continental, marine data and specified as a global $2' \times 2'$ resolution grid of the anomaly of the total magnetic intensity (Maus et al., 2009). So the EMAG2 geomagnetic anomaly grid map with $2' \times 2'$ resolution (EMAG2, 2011) was utilized for the GGCAN study.

3.2. *GravAN and GeomAN Algorithm Based on MMAE*. Kalman filters are widely used in integrated navigation systems. In this paper a Kalman filter and MMAE are utilized to construct GravAN and GeomAN algorithms. Firstly, a confidence area was formed and the size was determined by the INS position Circular Error Probability (CEP) and Rayleigh Distribution; it can be seen in Figure 2. Then, Kalman filters were constructed at every regularly arranged grid in the confidence area. The measurements for Kalman measurement equations are composed from the differences between the measured anomaly and the anomalies of the filters. The system model for the Kalman filter can be provided as follows:

$$
X = \begin{pmatrix}
X_{1,k} \\
X_{2,k}
\end{pmatrix}
= \phi_{k-1,k} \begin{pmatrix}
X_{1,k-1} \\
X_{2,k-1}
\end{pmatrix}
+ G_{k-1} \begin{pmatrix}
W_{1,k-1} \\
W_{2,k-1}
\end{pmatrix}
$$

(1)

$$
Z = \begin{pmatrix}
Z_{1,k} \\
Z_{2,k}
\end{pmatrix}
= H_k \begin{pmatrix}
X_{1,k} \\
X_{2,k}
\end{pmatrix}
+ \begin{pmatrix}
V_{1,k} \\
V_{2,k}
\end{pmatrix}
$$

(2)

Figure 2. Sketch Map of Confidence Interval and Validate Location.
The subscripts 1 and 2 indicate GravAN and GeomAN respectively in the whole paper, where:

\( X \) – Actual value of gravity/geomagnetic anomaly and it is treated as the state vector.

\( Z \) – The gravity/geomagnetic measurement vector.

\( W \) – System noise. \( W_{1,k} \sim N(0,q_1) \) and \( W_{2,k} \sim N(0,q_2) \), \( Q = \text{diag}(q_1,q_2) \).

\( V \) – Measurement noise. \( V_{1,k} \sim N(0,r_1) \) and \( V_{2,k} \sim N(0,r_2) \), \( R = \text{diag}(r_1,r_2) \).

The noises are assumed to be uncorrelated with each other. Five Kalman filter equations are listed as follows:

\[
\begin{align*}
X_k^- &= \phi X_{k-1} \\
P_k^- &= \phi P_{k-1} \phi^T + Q \\
K_k &= P_k^- H^T / (HP_k^- H^T + R) \\
X_k &= X_k^- + K_k (Z_k - HX_k^-) \\
P_k &= (I - K_k H) P_k^-
\end{align*}
\] (3)

Kalman filter computations are executed in every filter individually. Thereafter, the filtering residuals are defined as:

\[
\begin{pmatrix}
\delta_{1,k+1} \\
\delta_{2,k+1}
\end{pmatrix} =
\begin{pmatrix}
Z_{1,k+1} - X_{1,k+1|k} \\
Z_{2,k+1} - X_{2,k+1|k}
\end{pmatrix}
\] (4)

where:

\( \delta_{1,k+1} \sim N(0,P_{1,k|k-1} + r_1) \) and \( \delta_{2,k+1} \sim N(0,P_{2,k|k-1} + r_2) \) respectively.

\( P_{1,k|k-1} \) and \( P_{2,k|k-1} \) are the predicted covariance of the states of gravity and geomagnetic respectively.

The weighted residual and its smoothness value are defined (Hollowell, 1990) as:

\[
\begin{pmatrix}
WRS_{1,k+1} \\
WRS_{2,k+1}
\end{pmatrix} =
\begin{pmatrix}
(\delta_{1,k+1})^2 / (P_{1,k+1|k} + r_1) \\
(\delta_{2,k+1})^2 / (P_{2,k+1|k} + r_2)
\end{pmatrix}
\] (5)

\[
\begin{pmatrix}
SWRS_{1,k+1} \\
SWRS_{2,k+1}
\end{pmatrix} =
\begin{pmatrix}
\alpha \cdot WRS_{1,k+1} + (1 - \alpha)SWRS_{1,k} \\
\alpha \cdot WRS_{2,k+1} + (1 - \alpha)SWRS_{2,k}
\end{pmatrix}
\] (6)

where;

\( SWRS_{1,0} = SWRS_{2,0} = 1.0 \)

\( \alpha \) is smooth factor.

\( \alpha = 1.938p \) when \( p \) indicates the draft of the INS.

If \( SWRS \) is smaller, better position results can be achieved. So the smallest values of \( SWRS \) are sought, and the corresponding positions are regarded as the estimated results of positions for GravAN and GeomAN.
As the positions have been estimated, the following criterion is adopted to judge the reliability of the results:

\[
H = \frac{SWRS^*_{\text{min}} - SWRS_{1/2}^{\text{min}}}{SWRS_{1/2}^{\text{min}}} > H_t
\]  

(7)

where:

- $H$ is the reliability.
- $H_t$ is the threshold.
- $SWRS_{1/2}^{\text{min}}$ is the smallest $SWRS$ value, and $SWRS^*_{\text{min}}$ is the second smallest $SWRS$ value in the region around the corresponding position of $SWRS_{1/2}^{\text{min}}$. For estimating the results of position for GravAN and GeomAN, the region is defined by nine grids, including the corresponding position of $SWRS_{1/2}^{\text{min}}$ and eight surrounding grids, and is illustrated in Figure 3. From Equation (7) it can be seen that the reliability $H$ would increase if the differences between $SWRS^*_{\text{min}}$ and $SWRS_{1/2}^{\text{min}}$ increased. When the reliability $H$ is larger than threshold $H_t$, the position results would be considered as valid, otherwise they would be invalid.

As the position results of $SWRS_{1/2}^{\text{min}}$ are confirmed as valid, more accurate estimation of positions can be calculated with weighted methods. Equations (8) and (9) are presented here for position and weighted
calculation respectively:
\[
\begin{align*}
\lambda_1 &= \sum_{i=1}^{9} (W_i^1 \cdot \lambda_i^1) \\
\varphi_1 &= \sum_{i=1}^{9} (W_i^1 \cdot \varphi_i^1) \\
\lambda_2 &= \sum_{i=1}^{9} (W_i^2 \cdot \lambda_i^2) \\
\varphi_2 &= \sum_{i=1}^{9} (W_i^2 \cdot \varphi_i^2)
\end{align*}
\]
(8)

\[
\begin{align*}
W_i^1 &= \frac{\exp(-0.5 \times N_i^1)}{\sum_{i=1}^{9} \exp(-0.5 \times N_i^1)}, N_i^1 = \frac{SWRS_i^1}{SWRS_i^{min}}, 1 \leq i \leq 9 \\
W_i^2 &= \frac{\exp(-0.5 \times N_i^2)}{\sum_{i=1}^{9} \exp(-0.5 \times N_i^2)}, N_i^2 = \frac{SWRS_i^2}{SWRS_i^{min}}, 1 \leq i \leq 9
\end{align*}
\]
(9)

Where \((\lambda, \varphi)\) are the coordinates of the position, the weights are calculated with \(SWRS\) values at every grid. In Figure 3 point \(P_1(\lambda_1, \varphi_1)\) and \(P_2(\lambda_2, \varphi_2)\) are the more accurate estimation results.

4. OPTIMAL WEIGHT ALLOCATION PRINCIPLE (OWAP).

4.1. OWAP Algorithms. As the position results of GravAN and GeomAN have been obtained, OWAP is adopted to combine the gravity and geomagnetism aided navigation.

It is shown in Figure 3 that the position results of GravAN and GeomAN are \(P_1(\lambda_1, \varphi_1)\) and \(P_2(\lambda_2, \varphi_2)\), respectively. They are affected by various errors including measurement error, database error and the algorithm error. All the noises and errors are assumed to obey the law of Gaussian distribution (Ling et al., 1998):

\[
p(P_1) = N(\mu_1, \sigma_1^2)
\]
(10)

\[
p(P_2) = N(\mu_2, \sigma_2^2)
\]
(11)

After the following transformation, \(z_1\) and \(z_2\) obey the standard normal distribution:

\[
z_1 = \frac{P_1 - \mu_1}{\sigma_1}
\]
(12)

\[
z_2 = \frac{P_2 - \mu_2}{\sigma_2}
\]
(13)

The weighted average method of the gravity and geomagnetism combined aided navigation (GGCAN) is then constructed as:

\[
p = WP = [w_1, w_2][P_1, P_2]^T
\]
(14)

where:

\(W = [w_1, w_2]\) is the weight vector.

\(w_1\) and \(w_2\) are the weights of GravAN and GeomAN respectively.

\(P = [P_1, P_2]^T\).
Multiple random vectors \( Z \) obey the standard normal distribution after the transformation below:

\[
Z = A(P - U)
\]  

(15)

where:

\[
Z = [z_1, z_2].
\]

\[
A = \text{diag} \left[ \frac{1}{\sigma_1}, \frac{1}{\sigma_2} \right].
\]

\[
U = [\mu_1, \mu_2]^T.
\]

From Equation (15) we obtain:

\[
P = A^{-1}Z + U
\]  

(16)

Substituting Equation (16) into Equation (14):

\[
p = WP = W(A^{-1}Z + U) = WU + WA^{-1}Z
\]  

(17)

According to the theory of n-dimensional multivariate statistics, the probability density function of \( p \) is:

\[
f(p) = (2\pi)^{\frac{n}{2}} |WP^{-1}(P^{-1})'W'| \exp \left\{ \frac{1}{2} (p - WU)(WP^{-1}(P^{-1})'W'(p - WU))' \right\}
\]

\[
= (2\pi)^{-\frac{n}{2}} \sum_{i=1}^{n} w_i^2 \sigma_i^2 \exp \left\{ -\frac{1}{2} \sum_{i=1}^{n} w_i^2 \sigma_i^2 \left( y - \sum_{i=1}^{n} w_i^2 \sigma_i^2 \right)^2 \right\}
\]

(18)

This indicates that \( p \) obey normal distribution. In other words:

\[
p = N\left( \sum_{i=1}^{n} w_i \mu_i, \sum_{i=1}^{n} w_i^2 \sigma_i^2 \right)
\]

(19)

That is to say, the mathematical expectation of position result after fusing is the weighted expectation of GravAN and GeomAN. And the precision of GGCAN is:

\[
\sigma_p = \sqrt{\sum_{i=1}^{n} w_i^2 \sigma_i^2}
\]

(20)

Apparently, if \( \sigma_i \) are given, \( \sigma_p \) is only influenced and determined by \( w_i \). As the minimum of \( \sigma_p \) indicates the optimal accuracy of GGCAN. A mathematical model is constructed as follows:

\[
f(w_1, w_2) = \sum_{i=1}^{2} w_i^2 \sigma_i^2
\]

(21)

\[
\sum_{i=1}^{2} w_i = 1 (w_i \geq 0, i = 1, 2), \sigma_i (i = 1, 2)
\]

(22)
So the optimal problem is transformed into finding the conditional extreme value of this function $f(w_1, w_2)$ (Ling et al., 2000). This problem can be solved by the Lagrange Multiplier method.

4.2. The Solution of OWAP. The Lagrange Multiplier conditional extreme value equation can be written as below:

$$F = \sum_{i=1}^{2} w_i^2 \sigma_i^2 + \beta \left( \sum_{i=1}^{2} w_i - 1 \right)$$

(23)

where $\beta$ is the Lagrange Multiplier.

The partial differential of the $F$ with respect to $w_i (i=1,2)$ is assigned to 0, so a set of equations can be written as:

$$\begin{align*}
\frac{\partial F}{\partial w_1} &= 2w_1 \sigma_1^2 + \beta = 0 \\
\frac{\partial F}{\partial w_2} &= 2w_2 \sigma_2^2 + \beta = 0 \\
w_1 + w_2 &= 1
\end{align*}$$

(24)

Then the weight of GravAN and GeomAN could be obtained:

$$w_i = \frac{1}{\sigma_i^2 \sum_{i=1}^{2} \frac{1}{\sigma_i^2}}$$

(25)

Substituting Equation (25) into (20) the precision of GGCAN is:

$$\sigma_p = \sqrt{\frac{1}{\sum_{i=1}^{2} w_i^2 \sigma_i^2 \sum_{i=1}^{2} \frac{1}{\sigma_i^2}}}$$

(26)

4.3. Precision Analyses of OWAP. Let the precisions of GravAN and GeomAN be $\sigma_1$ and $\sigma_2$, respectively. Assuming that $\sigma_1 \leq \sigma_2$, the following inequality can be obtained:

$$\sigma_p = \frac{1}{\sqrt{\frac{1}{\sigma_1^2} + \frac{1}{\sigma_2^2}}} \leq \frac{1}{\sqrt{\frac{1}{\sigma_1^2}}} \leq \frac{1}{\sqrt{\frac{1}{\sigma_2^2}}}$$

(27)

This inequality indicates that with OWAP the precision of GGCAN is improved with respect to either GravAN or GeomAN. This is the essential difference between OWAP and other fusion methods. So more precise position results can be gained with the combination of gravity and geomagnetism aided navigation after OWAP.

5. GGCAN. The flow chart of the GGCAN algorithm is shown in Figure 4. With the MMAE-based GravAN and GeomAN algorithms mentioned in Section 3, and OWAP method mentioned in Section 4, the GGCAN algorithm can be provided. The fusing algorithm OWAP only works when both GravAN and GeomAN are valid.
As the criterion function, $H$ in Equation (7) can, to a certain extent, be regarded as the precision of aided navigation. It is assumed that:

$$
\begin{align*}
\sigma_1 &= \frac{SWRS_1^* - SWRS_1^{min}}{SWRS_1^{min}} \\
\sigma_2 &= \frac{SWRS_2^* - SWRS_2^{min}}{SWRS_2^{min}}
\end{align*}
$$

Thereafter, the weight matrix is formed according to Equation (25). With the weights the position result $P_1(\lambda_1, \varphi_1)$ of GravAN and the position result $P_2(\lambda_2, \varphi_2)$ of GeomAN are taken to calculate the final GGCAN position result which is marked with a red dot in Figure 3.

The final GGCAN position result can be obtained from the following equation:

$$
\lambda = w_1 \cdot \lambda_1 + w_2 \cdot \lambda_2, \varphi = w_1 \cdot \varphi_1 + w_2 \cdot \varphi_2
$$

6. SIMULATION RESULTS AND DISCUSSION. Gravity and geomagnetic anomaly maps for simulation are illustrated in Figure 5. Two special
Regions A and B in the West Pacific Ocean are selected, which are marked with black rectangles in Figure 5. The statistical information of gravity and geomagnetic anomalies in the two regions are collected and listed in Table 1. From this table it can be found that the standard deviation of the geomagnetic anomaly is bigger than the gravity anomaly in Region A, and smaller than the gravity anomaly in
Region B. So it perhaps means that there is more geomagnetic information in Region A and more gravity information in Region B.

Simulation parameters including initial position errors, INS drift, measurement errors, AUV velocity, smooth factor and reliability threshold $H_t$ are assigned; their values are listed in Table 2. The actual trajectories are simulated from South West to North East in Regions A and B. Under the same conditions four methods, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP) were used to provide position information in Regions A and B. The GGCAN(EW) was tested to prove the performance of OWAP. The trajectories of four aided navigation methods are showed in Figure 6. The position errors of these methods are showed in Figure 7.

It can be seen from Figures 6 and 7 that INS position errors increase with time. With the four methods, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP), the INS position errors can be corrected effectively by different amounts. The position errors of GGCAN(OWAP) were marked with red pentagrams in Figure 7 and, in these diagrams, GGCAN(OWAP) was lower and more steady than the three other methods.

For further quantitative analysis, the statistical parameters about navigation results are counted and listed in Table 3.

### Table 1. Gravity and geomagnetic parameters in the simulation regions.

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Longitude range (deg)</td>
<td>130–135</td>
</tr>
<tr>
<td>Latitude range (deg)</td>
<td>23–28</td>
</tr>
<tr>
<td>Grid interval (min)</td>
<td>2</td>
</tr>
<tr>
<td>Gravity anomaly (mgal)</td>
<td>Minimum -166·32</td>
</tr>
<tr>
<td></td>
<td>Maximum 134·98</td>
</tr>
<tr>
<td></td>
<td>Standard deviation 34·57</td>
</tr>
<tr>
<td>Geomagnetic anomaly (nT)</td>
<td>Minimum -279·87</td>
</tr>
<tr>
<td></td>
<td>Maximum 387·23</td>
</tr>
<tr>
<td></td>
<td>Standard deviation 88·02</td>
</tr>
</tbody>
</table>

### Table 2. Simulation parameters.

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Initial North location error (deg)</td>
<td>0·003</td>
</tr>
<tr>
<td>Initial East location error (deg)</td>
<td>0·003</td>
</tr>
<tr>
<td>INS North drift (deg/h)</td>
<td>0·48</td>
</tr>
<tr>
<td>INS East drift (deg/h)</td>
<td>0·64</td>
</tr>
<tr>
<td>Gravimeter measure noise (mgal²)</td>
<td>10</td>
</tr>
<tr>
<td>Magnetometer measure noise (nT²)</td>
<td>10</td>
</tr>
<tr>
<td>AUV velocity (n mile/h)</td>
<td>20</td>
</tr>
<tr>
<td>Smooth factor</td>
<td>0·969</td>
</tr>
<tr>
<td>$H_t$</td>
<td>0·4</td>
</tr>
</tbody>
</table>

Region B. So it perhaps means that there is more geomagnetic information in Region A and more gravity information in Region B.

Simulation parameters including initial position errors, INS drift, measurement errors, AUV velocity, smooth factor and reliability threshold $H_t$ are assigned; their values are listed in Table 2. The actual trajectories are simulated from South West to North East in Regions A and B. Under the same conditions four methods, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP) were used to provide position information in Regions A and B. The GGCAN(EW) was tested to prove the performance of OWAP. The trajectories of four aided navigation methods are showed in Figure 6. The position errors of these methods are showed in Figure 7.

It can be seen from Figures 6 and 7 that INS position errors increase with time. With the four methods, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP), the INS position errors can be corrected effectively by different amounts. The position errors of GGCAN(OWAP) were marked with red pentagrams in Figure 7 and, in these diagrams, GGCAN(OWAP) was lower and more steady than the three other methods.

For further quantitative analysis, the statistical parameters about navigation results are counted and listed in Table 3. In this table it can be seen that in Region A the performance of GeomAN is better than GravAN and in Region B this comparison is opposite. So, as mentioned before, it is apparent that there is more geomagnetic
Table 3. Simulation results of four types of aided navigation systems.

<table>
<thead>
<tr>
<th>Measure times</th>
<th>GRAVAN</th>
<th>GEOFAN</th>
<th>GGCAN (EW)</th>
<th>GGCAN (OWAP)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Validate times</td>
<td>213</td>
<td>112</td>
<td>169</td>
<td>169</td>
</tr>
<tr>
<td>Success rate</td>
<td>52.58%</td>
<td>85.92%</td>
<td>79.34%</td>
<td>69.48%</td>
</tr>
<tr>
<td>Error (n mile)</td>
<td>Mean</td>
<td>1.81</td>
<td>1.73</td>
<td>1.53</td>
</tr>
<tr>
<td></td>
<td>variance</td>
<td>0.95</td>
<td>0.85</td>
<td>0.90</td>
</tr>
</tbody>
</table>

6a). True, INS, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP) Trajectories in Region A.

6b). True, INS, GravAN, GeomAN, GGCAN(EW) and GGCAN(OWAP) Trajectories in Region B.

Figure 6. The Trajectories of the Two Simulation Tests in Region A and B.
information than gravity in Region A, and more gravity information than geomagnetic in Region B.

In the four aided navigation methods, GGCAN(EW) and GGCAN(OWAP) have a distinct advantage in success rate and position error than GravAN and GeomAN, in either Region A or B. In Region A the success rate of GGCAN(OWAP) is 91.55%,
much higher than 52.58% of GravAN and 85.92% of GeomAN. Similarly, in Region B the success rates are 93.43%, 79.34% and 69.48%, respectively. This means that GGCAN(OWAP) can be valid when GravAN or GeomAN fails. In Region A the mean position error of GGCAN(OWAP) is 1.51 n. miles, smaller than 1.81 n. miles of GravAN and 1.73 n. miles of GeomAN. In Region B these errors are 1.34, 1.53 and 1.99 n. miles. The improvement in success rate and position accuracy show that the GGCAN(OWAP) method presented in this study is more accurate and robust than GravAN or GeomAN.

Moreover, in both Regions A and B, the success rate of GGCAN(EW) and GGCAN(OWAP) are equal, but the position errors of GGCAN(OWAP) are smaller. This indicates that the proposed weighted method of OWAP is more suitable for GGCAN than the equal weighted method.

7. CONCLUSIONS. Some aided navigation methods are necessary to correct the INS position error in underwater passive navigation. Gravity aided navigation and geomagnetic aided navigation are two suitable methods for establishing aided navigation systems on Autonomous Underwater Vehicles (AUVs). But in some regions, gravity or geomagnetic navigation may fail due to the inconspicuous gravity/geomagnetic variation and some other errors. In this paper, the Gravity-Geomagnetism Combined Aided Navigation (GGCAN) is proposed. In this method a Kalman filter and Multiple Model Adaptive Estimation (MMAE) are utilized to get the position results of Gravity Aided Navigation (GravAN) and Geomagnetic Aided Navigation (GeomAN). With the weighted method of Optimal Weight Allocation Principle (OWAP), the two results are weighted and combined. Two regions in the West Pacific Ocean were selected in the simulation and the improved success rate and position accuracy show that OWAP is more suitable for GGCAN than the equal weighted method, and GGCAN(OWAP) presented in this study is more accurate and robust than GravAN or GeomAN.

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